

Offshore Turbine Arrays: Numerical Modeling and Experimental Validation



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Motivation

- Upwind turbines affect downwind turbine: Performance, Loads, Wakes
- Economic consequences:**
 - Decrease in performance
 - Loss of energy production up to 20%
- Increased turbulence**
 - Fatigue loading on rotors
 - Varying loads on gearbox
 - Decreased life span
 - Increased O&M costs
- US DOE goal: Produce 20% of electric energy from wind by 2030
- Array experiments are performed at the UNH Flow Physics Facility
- Numerical model can be used to predict/validate results and decrease experimental efforts and costs



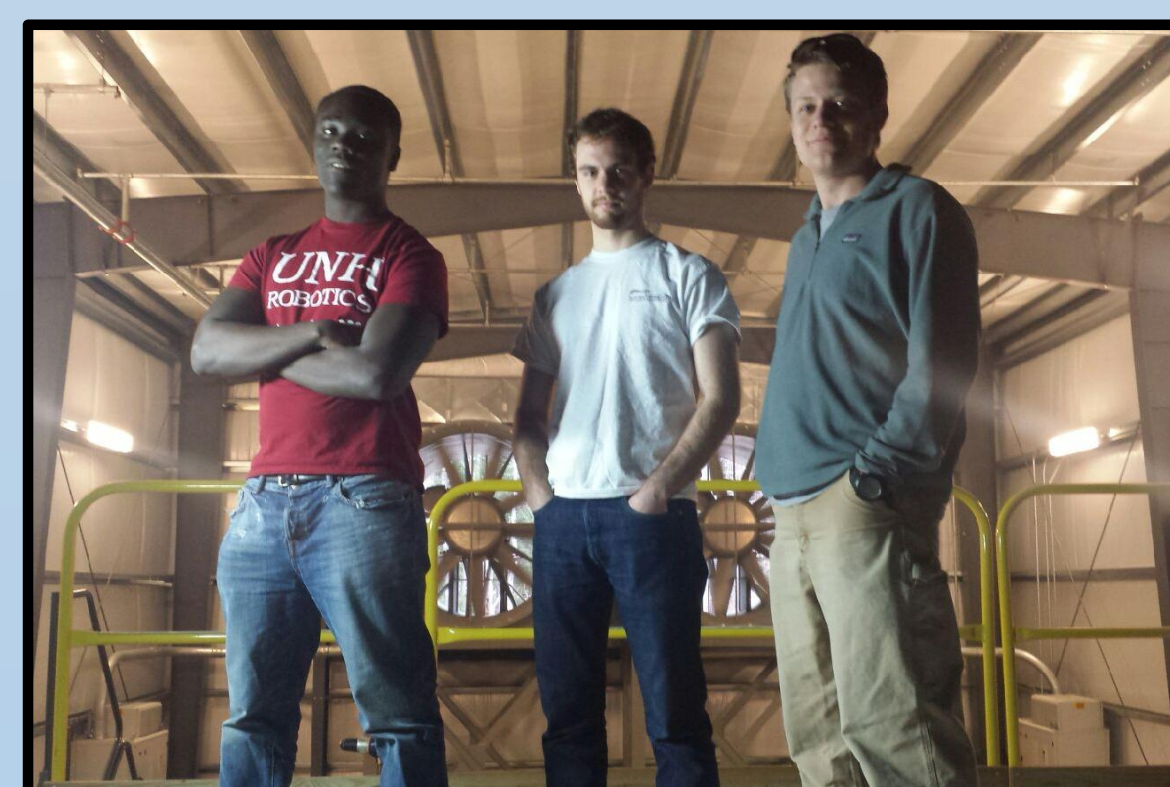
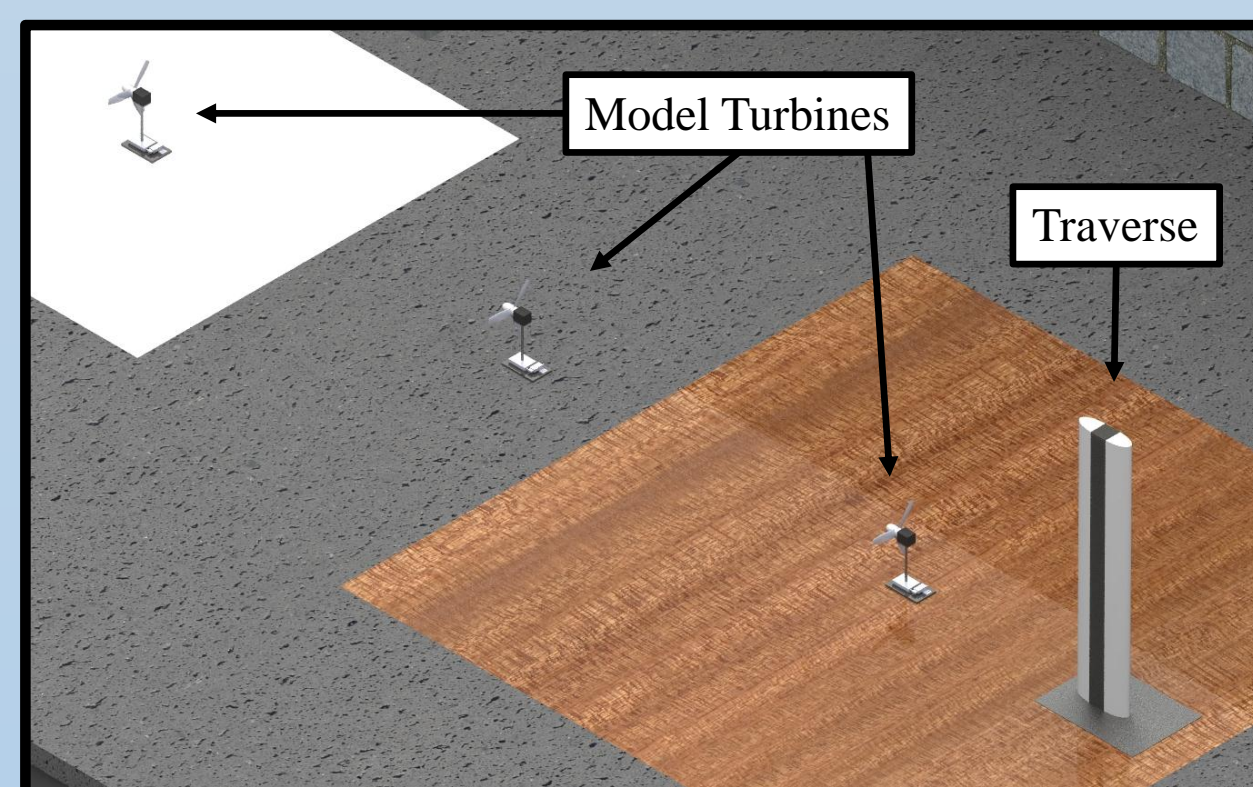
Horns Rev 1, Inset: Normalized power at each turbine in a central row at Horns Rev 1 (Barthelme et al., 2009)

Experimental Data

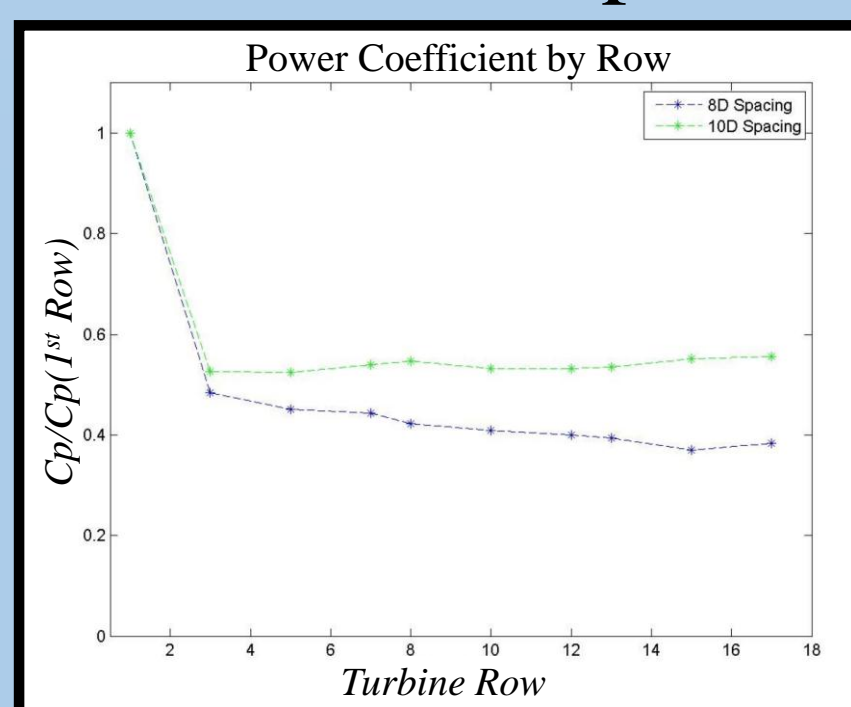
- Scale Model Turbines**
- Rotor diameter of 0.25 m
 - Hub height 0.75 D (1:500 scale)
 - NACA 2412 airfoils
 - Anaheim Automation DC motor
 - Force balance stand to measure drag force

- Porous Disks**
- Experimental tool, saves cost when modeling many turbines
 - Diameter of 0.25 m
 - Hub Height 0.75 D
 - Porosity of 0.48
 - Wake similarity 8D downstream

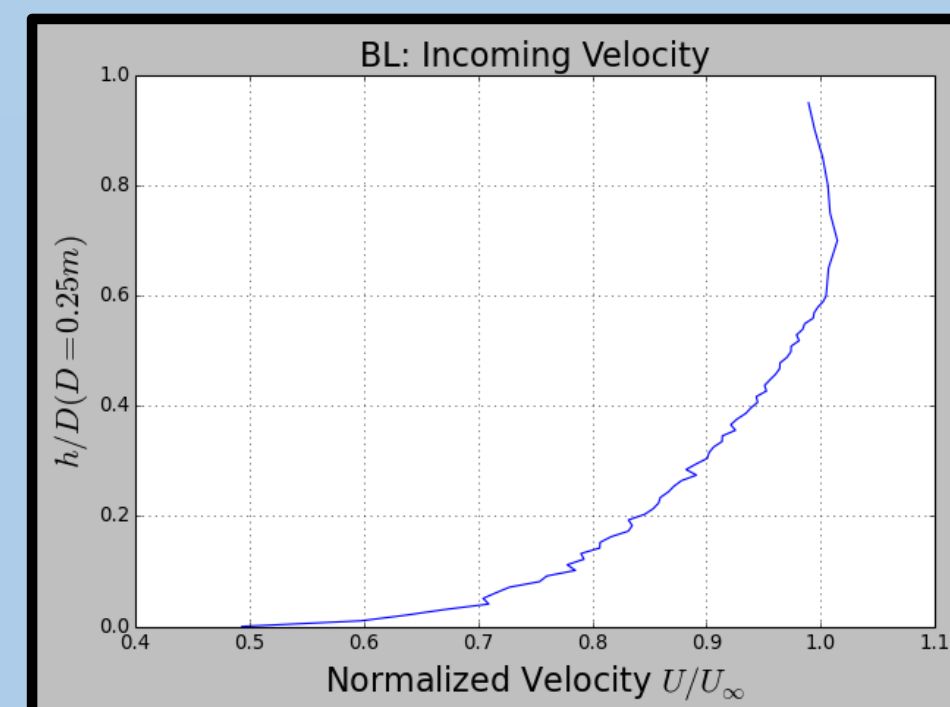
Data Collection



- Velocity and turbulence intensity measurements from hot-wire
 - Profiles of wind tunnel inlet and profiles behind turbine and porous disk
- Drag measurements taken with load cells
 - Coefficient of power for first and downstream turbines



Courtesy of John Turner V, UNH



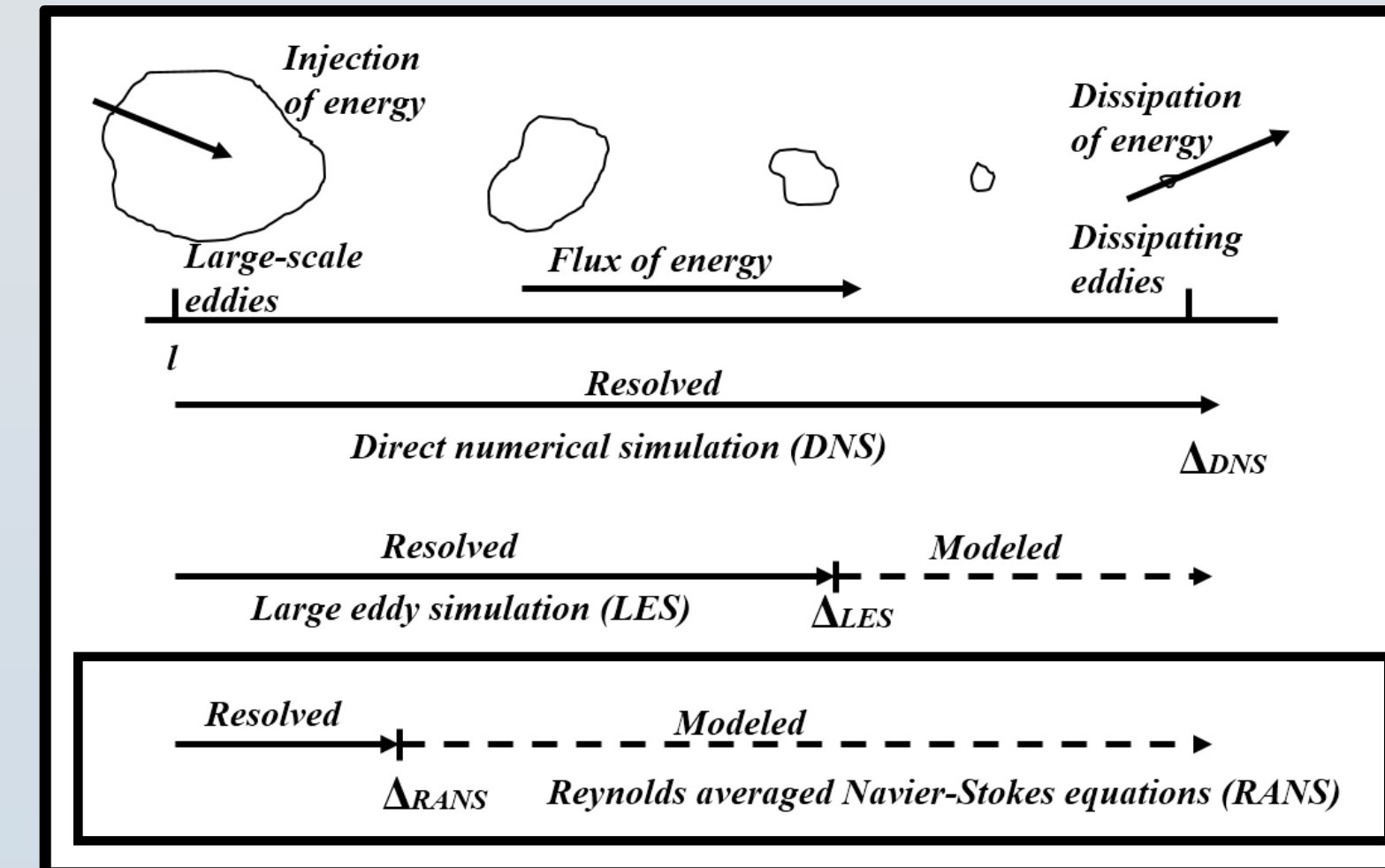
Computational Fluid Dynamics (CFD)

- Turbines modeled as actuator disks
- Turbulence modeling. Two equation model
- Flow in box → single turbine → arrays
- At each step, numerical results compared to experimental data
- Model parameters analyzed and modified

Varied Parameters	
k ,	Turbulence kinetic energy
l ,	Length scale
ε ,	Kinetic energy dissipation
I ,	Turbulence intensity
	Mesh refinement
	Boundary Conditions

RANS Modeling (k-ε)

- Reynold's Averaged Navier Stokes equations
- Effects of turbulence on the mean flow
- Model Reynolds stresses through transport equations for k and ϵ



RANS Equations:

$$\rightarrow \frac{\partial \bar{v}_i}{\partial x_i} = 0 \quad \text{Continuity Equation}$$

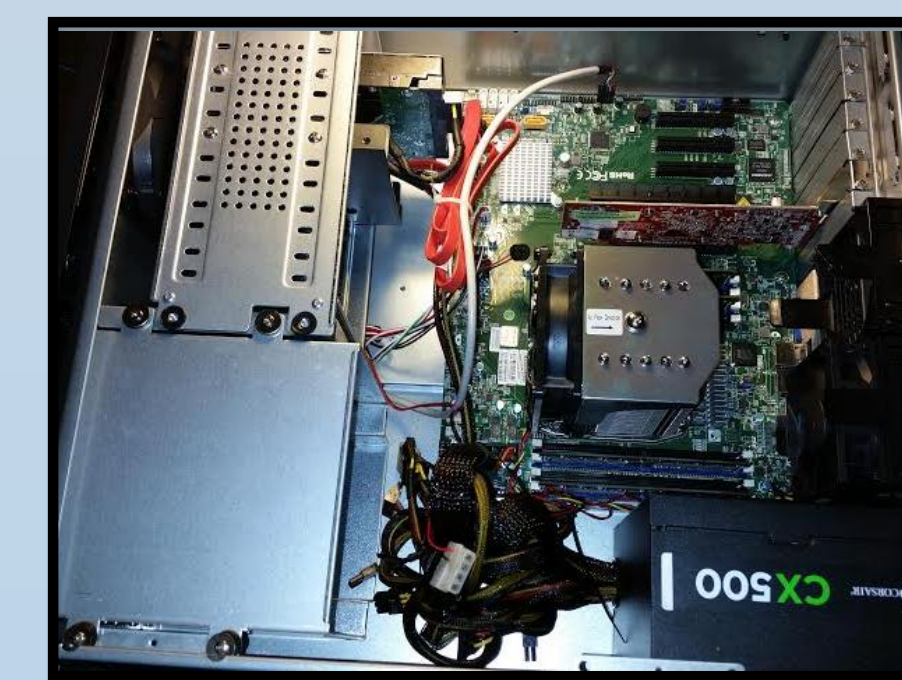
$$\rightarrow \rho \left(\frac{\partial \bar{v}_i}{\partial t} + \bar{v}_j \frac{\partial \bar{v}_i}{\partial x_j} \right) = - \frac{\partial \bar{p}}{\partial x_i} + \frac{\partial}{\partial x_j} (\bar{\tau}_{ij} - \rho \overline{v'_i v'_j}) \quad \text{Momentum Equation}$$

$$\rightarrow \frac{\partial (\rho k)}{\partial t} + \nabla \cdot (\rho k U) = \nabla \cdot \left[\frac{\mu_t}{\sigma_k} \nabla k \right] + 2 \mu_t S_{ij} * S_{ij} - \rho \epsilon \quad \text{Kinetic Energy Transport Equation}$$

$$\rightarrow \frac{\partial (\rho \epsilon)}{\partial t} + \nabla \cdot (\rho \epsilon U) = \nabla \cdot \left[\frac{\mu_t}{\sigma_\epsilon} \nabla \epsilon \right] + C_{1\epsilon} \left(\frac{\epsilon}{k} \right) 2 \mu_t S_{ij} * S_{ij} - C_{2\epsilon} \rho \left(\frac{\epsilon^2}{k} \right) \quad \text{Dissipation Equation}$$

Open-Sourced Engineering

- Ubuntu – Necessary for OpenFOAM
- OpenFOAM – CFD C++ Toolbox
- Paraview – Data viewer
- Python – Post processing
- Git – Version control system
- Github – Source code hosting and sharing

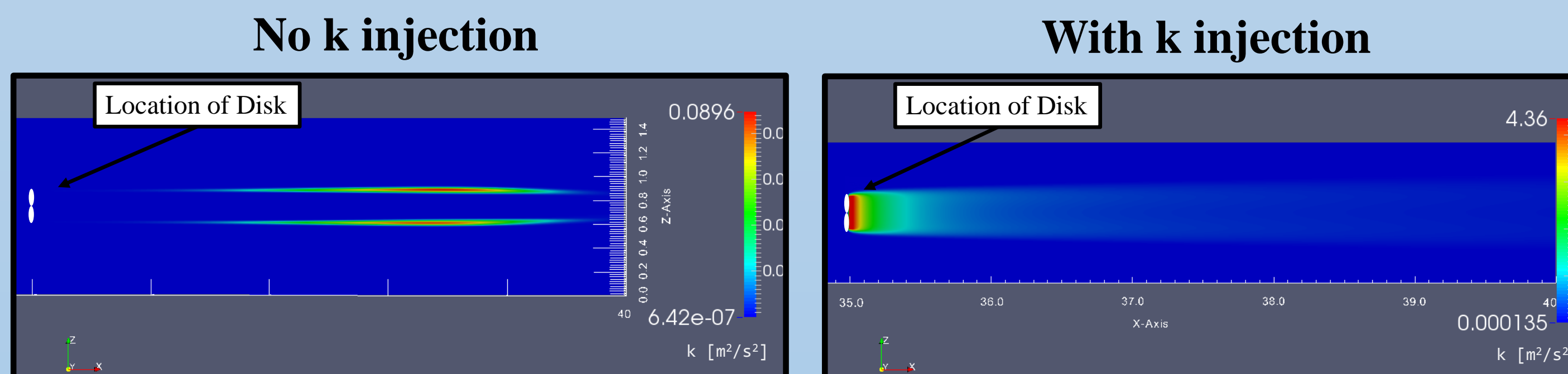


Simulations run on custom built server designed as head node of small future CFD cluster



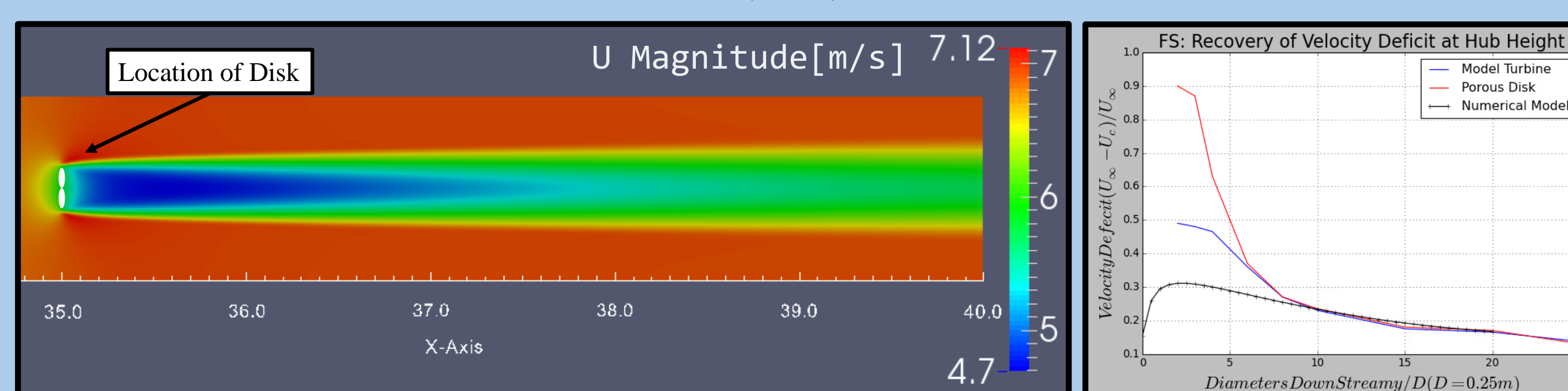
OpenFOAM Numerical Model

Turbulence Injection (k and ε):

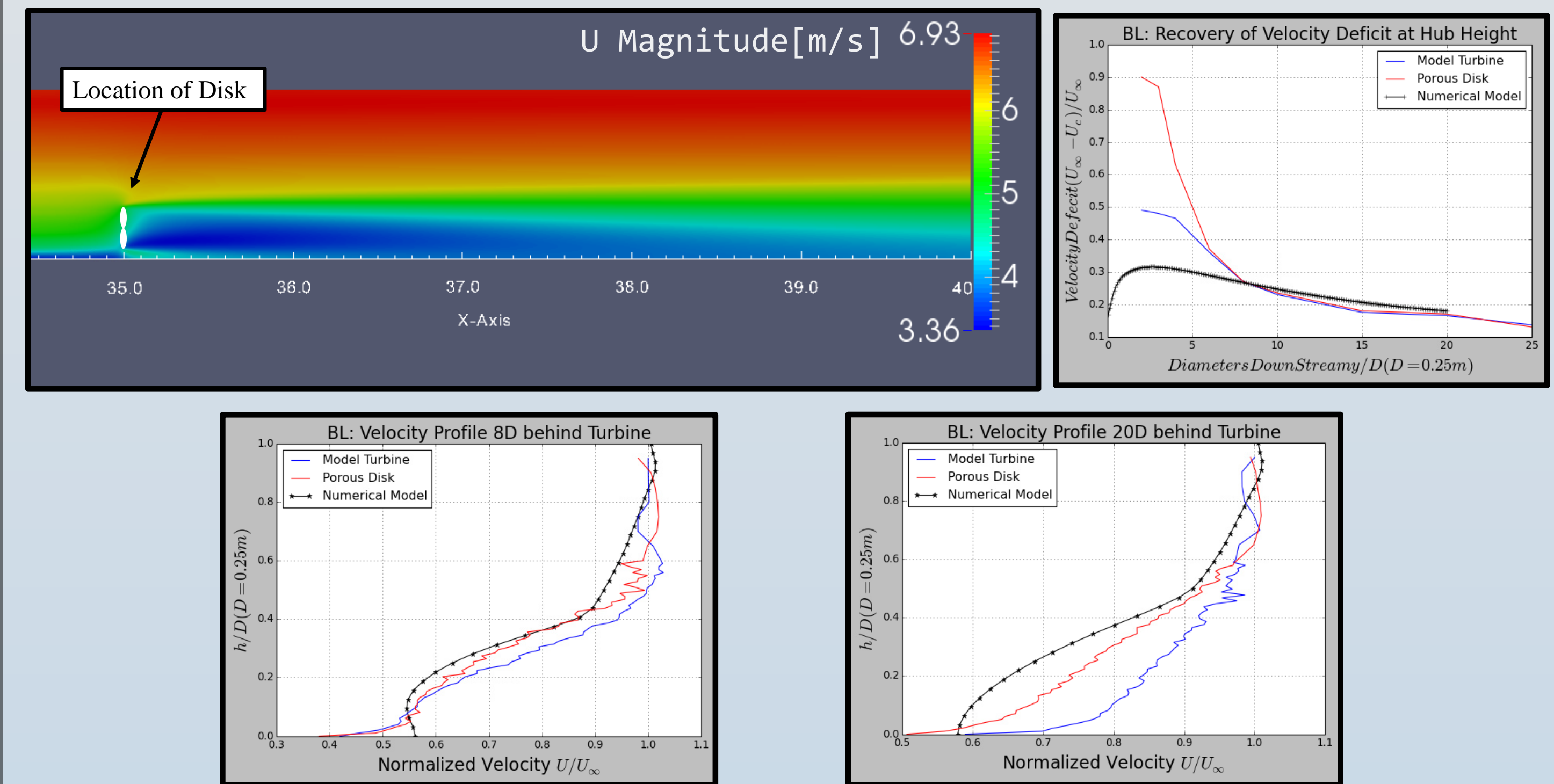


- Blades not resolved with actuator disks so less turbulence in wake
- Only turbulence due to shear from the velocity gradient at wake boundaries
- Velocity recovers too slowly due to decreased mixing
- Necessary to inject turbulence at the actuator disk

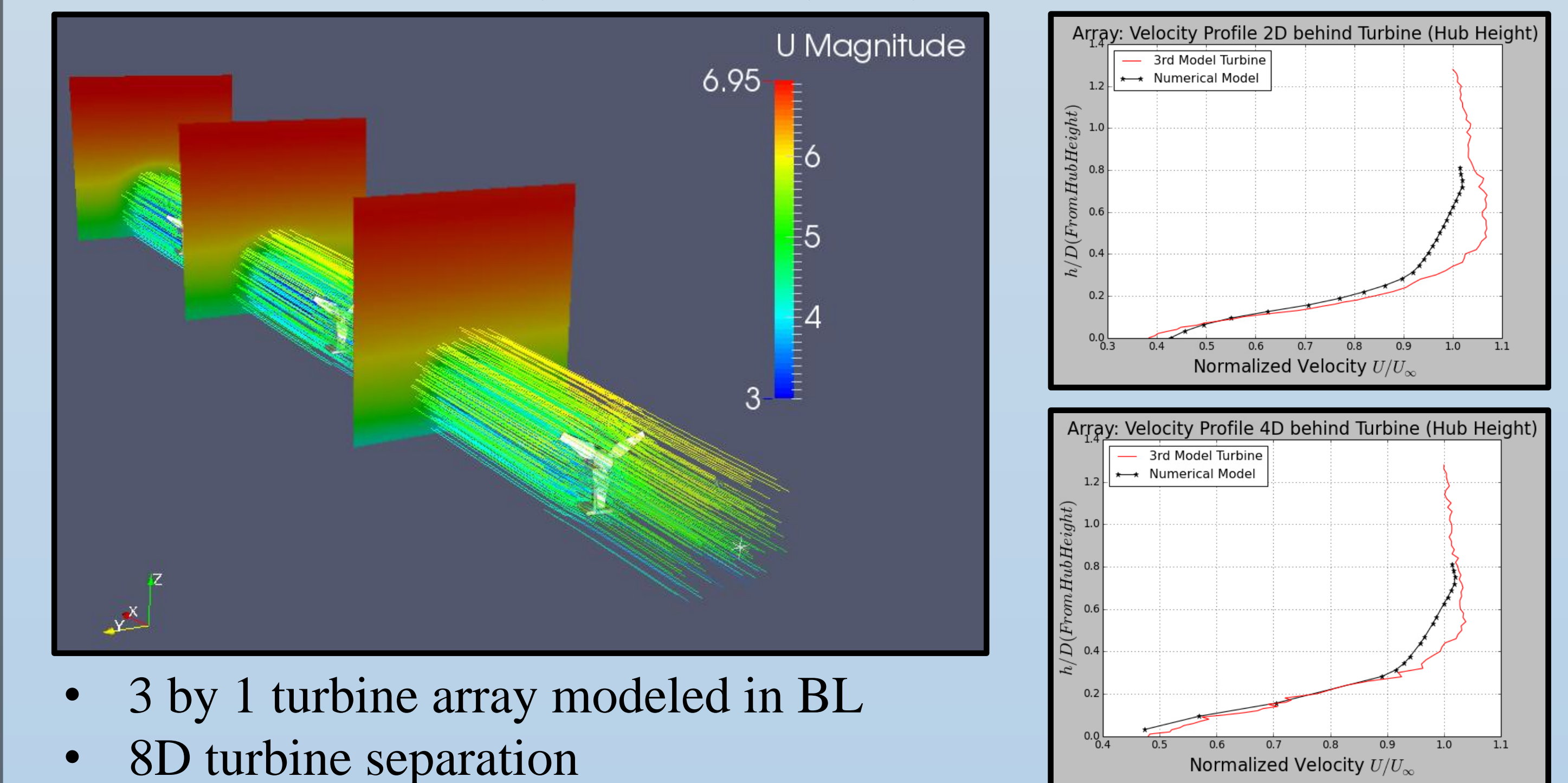
Actuator Disk: Freestream (FS)



Actuator Disk: Boundary Layer (BL)

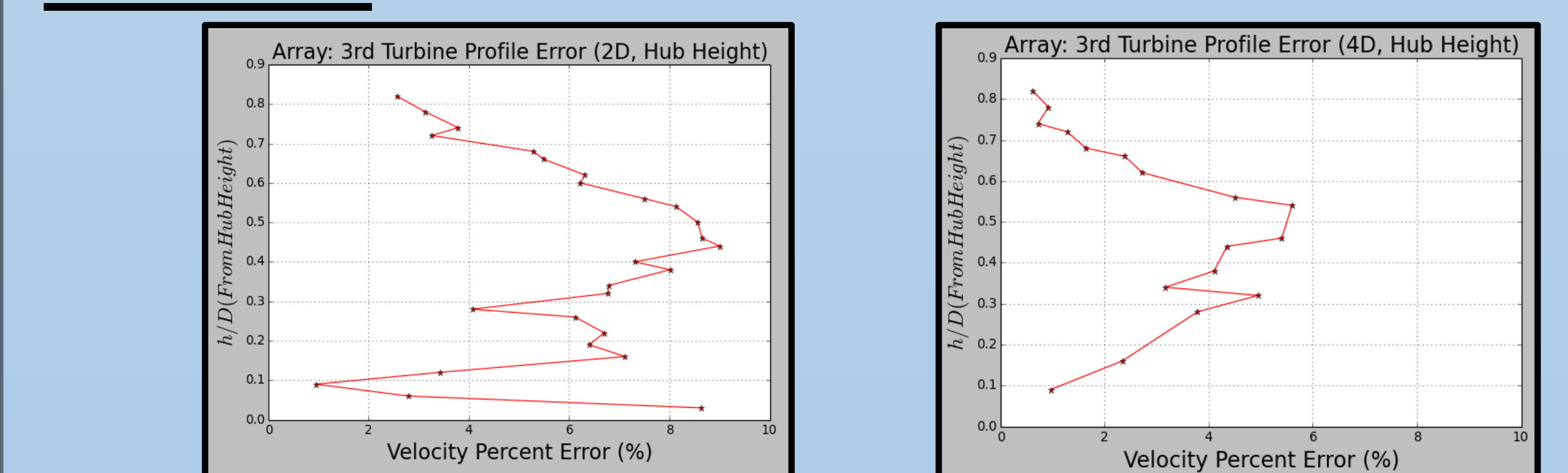


Actuator Disk Array: Boundary Layer



- 3 by 1 turbine array modeled in BL
- 8D turbine separation

Outcomes



- Percent difference for 3rd row turbine velocity profile
- Less than 10% at 2D
 - Less than 6% at 4D

Framework for studying turbine arrays was developed

Future Work

- Develop actuator line model for increased model accuracy
- Improve computational power of CFD cluster
- Apply model to larger arrays
- Algorithm for turbine array optimization

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