



Team Members: Matt Barricklow, Aaron Chapman, Kimia Fereydooni, Yanyuan He, Jon Lewis, Haoming Ni, Joe Roylance, Brennan Smith, Hongjie Zhang
Team advisor: Alireza Ebadi **Team Pilot:** Daren Hudson

SAE Aero Design East Overview

Design a plane that can deliver colonists and supplies safely to the surface of Mars

Plane Requirements:

- Maximum wingspan of 144 in
- Maximum gross weight of 55 lbs.
- Electric powered with 750-watt power limiter

Payload Requirements:

- Use unmodified water bottle and football (as supplies)
- Deliver ping-pong balls (as colonists) through Colonist Delivery Aircraft (CDA) with the following requirements:
 - Maximum weight of 9.0 oz.
 - At least two degrees of freedom must be controlled autonomously



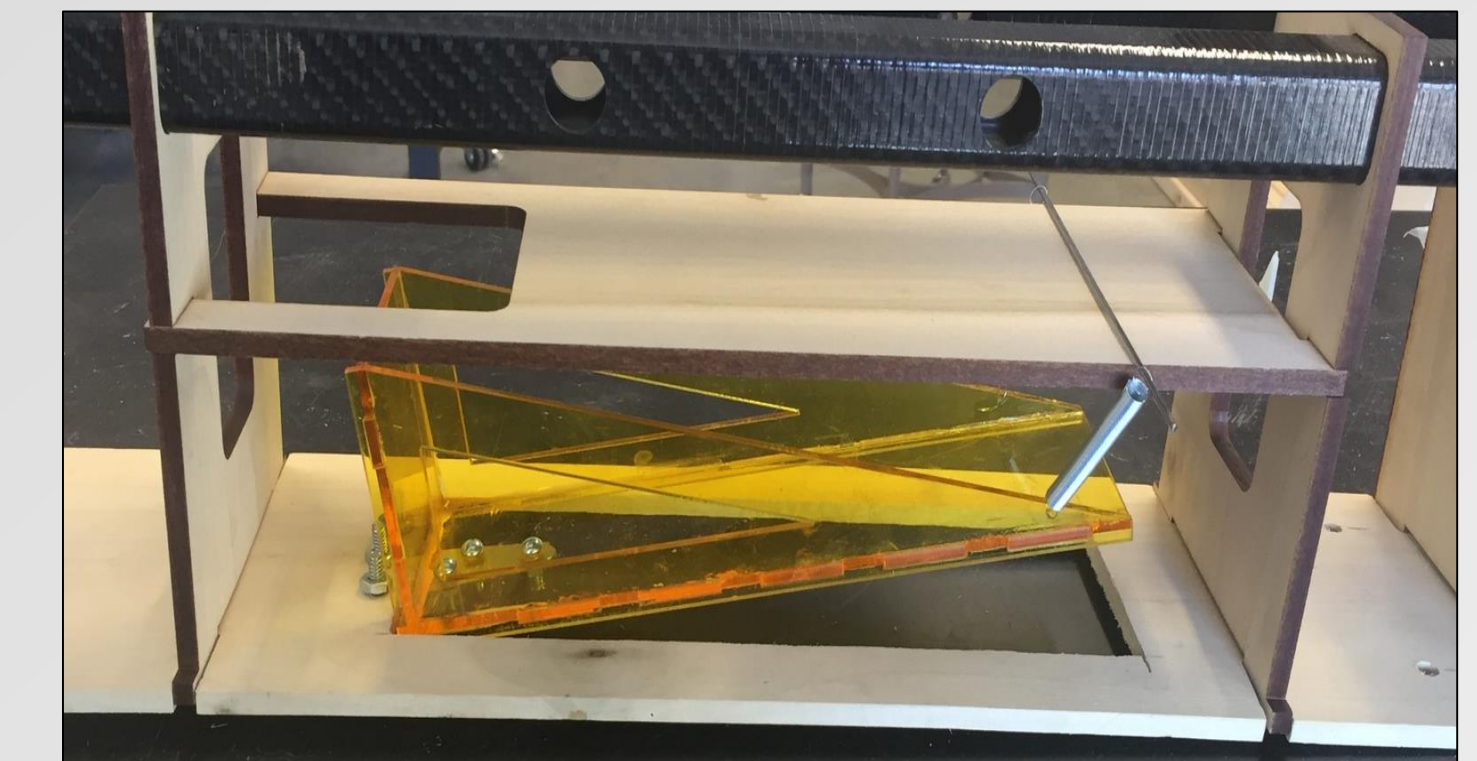
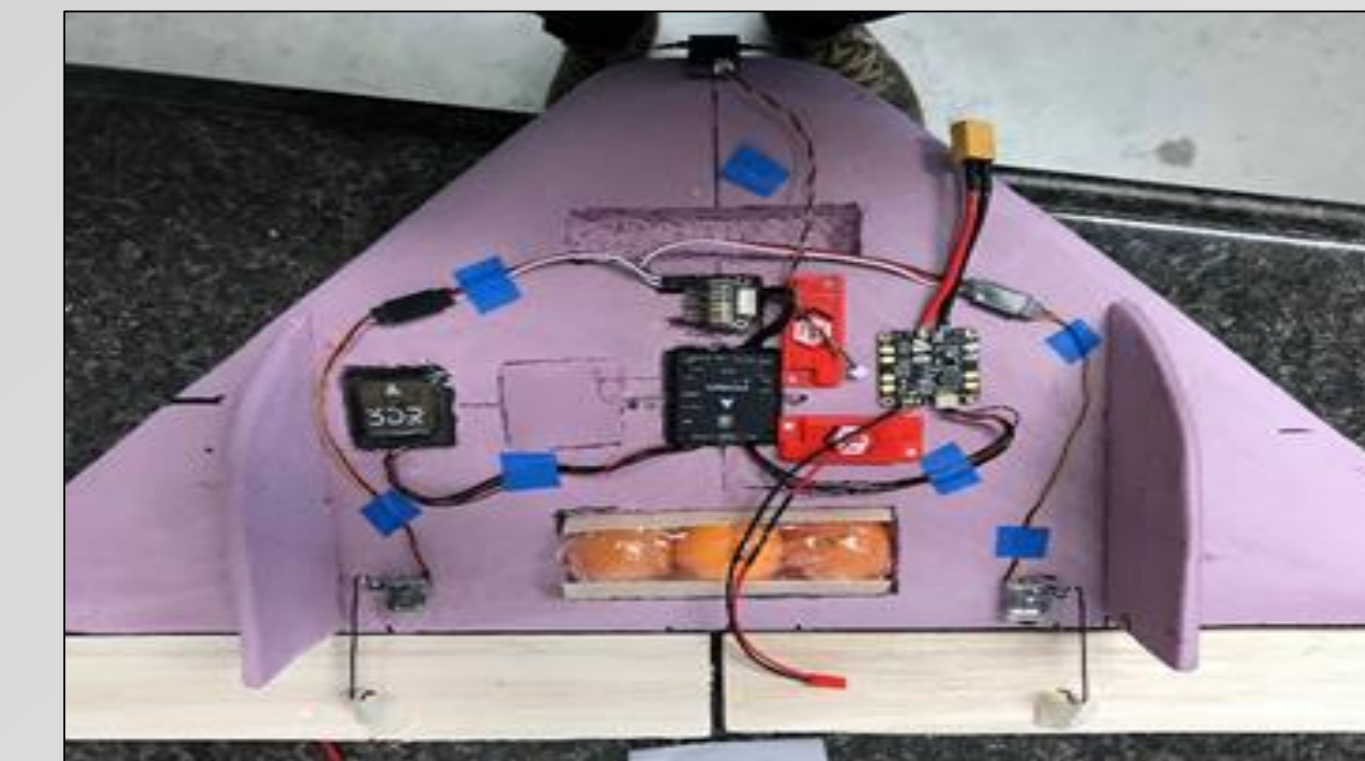
Glider and Payloads

Glider Design:

- Weight - 8.9 oz.
- Number of colonists – 3
- Flying wing shape
 - Wing taper: 2.3 in to 23 in
- 2 vertical stabilizers

Glider Controls:

- Pixhawk programmed for autonomous control
- Pitch and roll maneuverability
- GPS for landing location
- Radio transmitter for fail-safe



Payloads:

- One 16.9 fl oz. water bottle
 - Carried internally
 - Released via spring door
- Two footballs
 - Carried externally
 - Released via tension wire

Dropping Mechanism:

- Water bottle and footballs released simultaneously through an interconnected pin system
- Glider released separately via servo-pin system

Analysis

ANSYS Fluent:

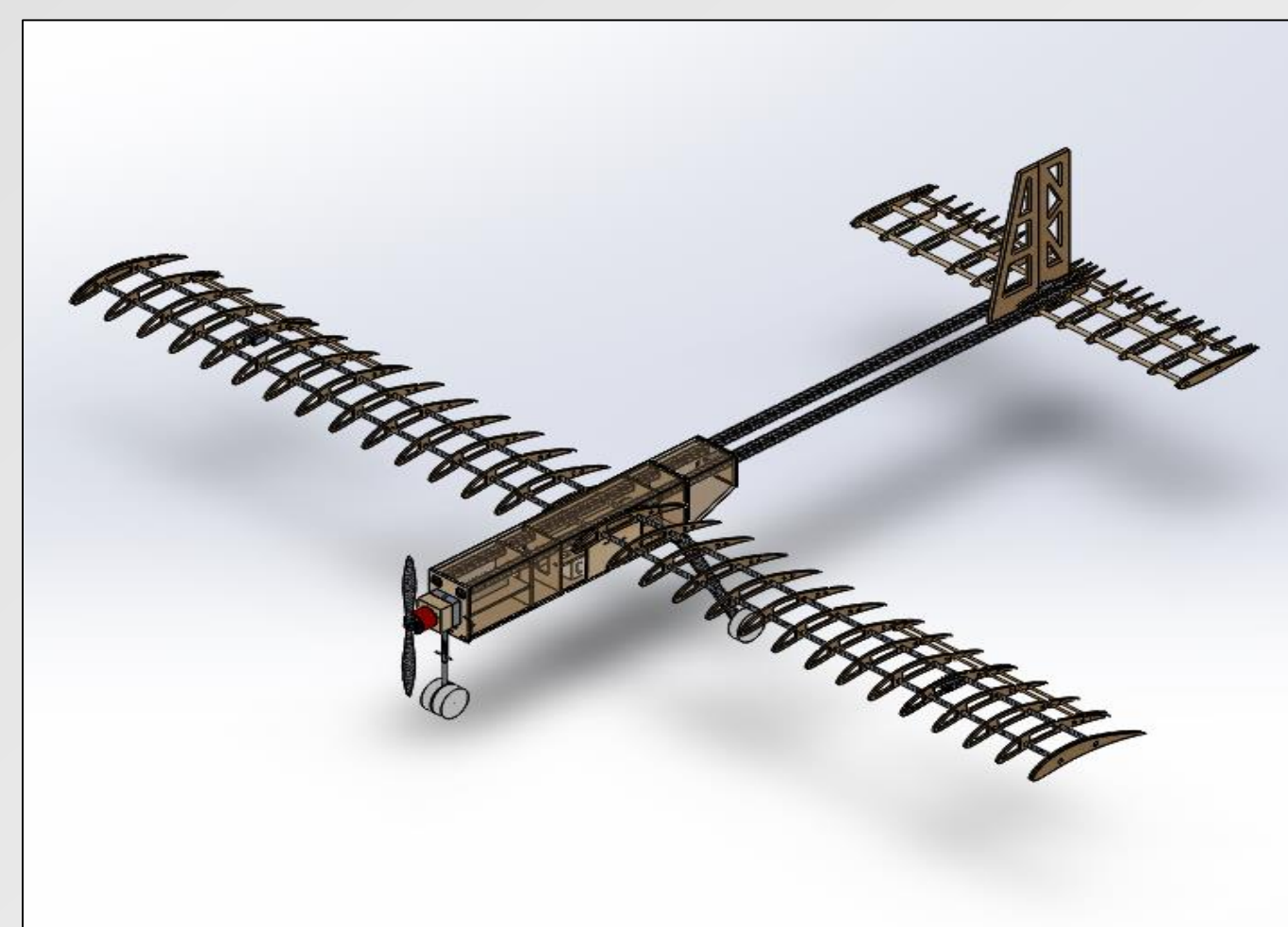
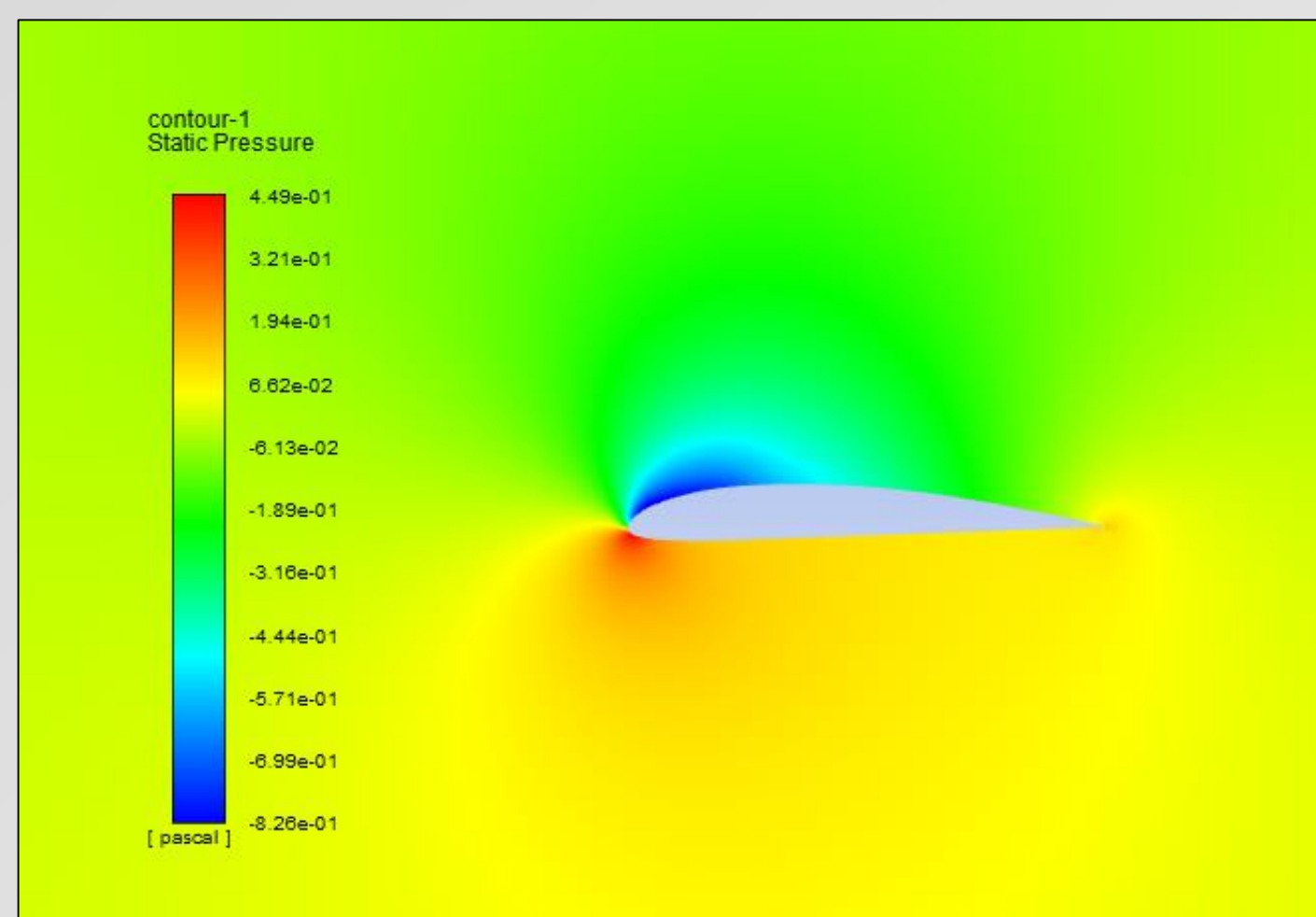
- 2D simulation of airflow around ClarkY, Eppler 423, S1223 airfoils
- Theoretical lift and drag coefficients for each airfoil at different angles of attack

SolidWorks:

- Stress analysis of plane during takeoff and landing
- CG location determination for plane and glider

MATLAB:

- Motor and propeller analysis
- Optimize number of dynamic payloads to maximize score



Controls

Flight Controls:

- Radio transmitter to control surface servos and motor

Mission Controls:

- Pixhawk for dropping mechanism control, altitude and direction readings
- First-person view camera
- In-flight data recorded in Mission Planner



Manufacturing

Laser Cutting

- Airfoil geometry (Balsa wood)
- Fuselage components (Poplar wood)
- Water bottle dropping mechanism (Acrylic sheet)

Machining

- Carbon fiber tubes holes
- Front landing gear

3D Printer

- Payloads' dropping mechanism
- Vortex generators



Test Flight and Resolutions

Plane Issues:

1. Front landing gear rigidity
2. Weak controllability between elevator and servo
3. Insufficient lift force
4. Dropping mechanism failure

Plane Solutions:

1. Reinforced welding spot and collar on landing gear
2. Replaced servo arms and horns
3. Changed propeller, added nose cone to motor
4. Adjusted payload tension wires

Glider Issues:

1. EPP foam provided insufficient strength
2. CG issues, tail heavy
3. Unstable control system
4. Failed to safely release from main plane

Glider Solutions:

1. Changed to insulation foam
2. Traditional glider shape changed to flying wing design
3. Adjusted PID controller gain, added vertical stabilizers
4. Removed tail skids from plane



Results:

Design report – 14th place
 Oral presentation – 12th place
 Flight score – 9th place
Overall – 15th place

Design and Dimensions

Main Wing:

- Eppler 423 Airfoil
 - Chord – 14.6 in
 - Aspect Ratio – 8.1

Fuselage:

- Length - 34.5 in
- Width – 5.0 in

Empennage:

- Horizontal stabilizer (NACA 0012)
 - Chord - 10.24 in
 - Span - 42.12 in
- Vertical stabilizer
 - Height - 17.5 in

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